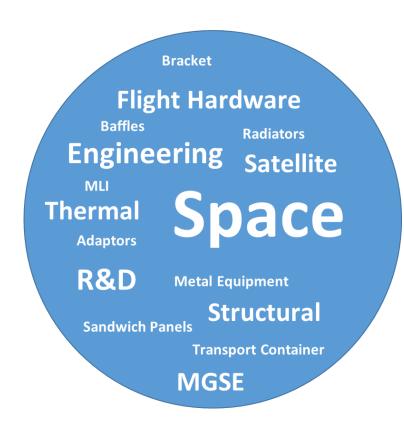


Company introduction

Admatis Ltd.

Bárczy Tamás



General



Foundation: 2000

Owners: Hungarian private persons

Location: Miskolc, Hungary

Size: SME

Position: prime in Hungary

Focus: space, thermal & structural

engineering, materials, aging,

navigation support, MGSE

Certificates: AS/EN9100, ISO9001, CCR reg, Ncage

code, ESA cost audit



Motto

"Whether you believe you can do a thing or not, you're right"
Henry Ford

We believe we can do it.

Customers / Partners



































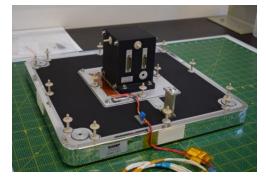
Competencies

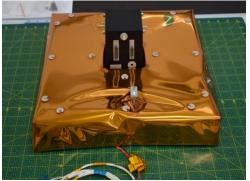


flight hardware manned missions

flight hardware unmanned mission

ground support equipments





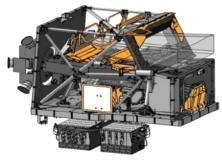
SPACE

Subprime in Hungary
Mechanical and Thermal
engineering
CAD design
FEM analysis
ESA documentation
Project and PA management
Product development

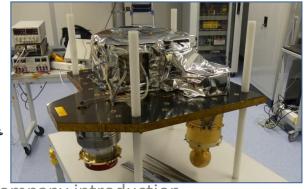
MATERIAL SCIENCE

Materials R&D
Aging
Thermo-optical coatings
Conversion coatings
Multi Layer Insulation, MLI

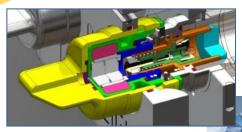






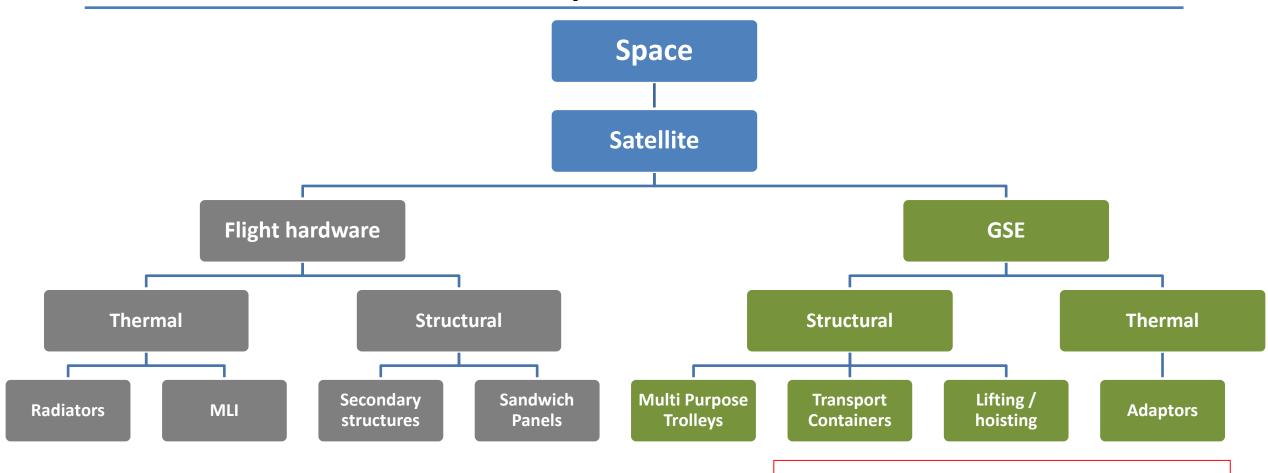








Competencies



NO IPR constraints.
All IPRs owned by ADMATIS.

Engineering

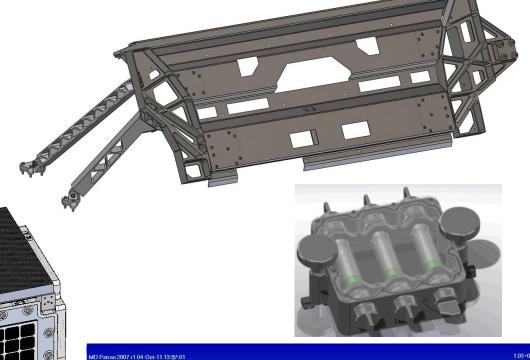


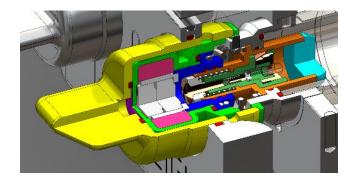
• Building Requirements Specification

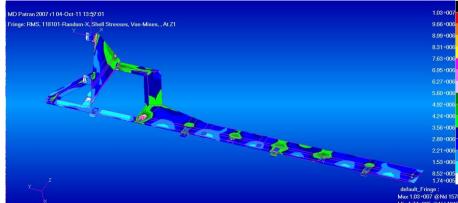
CAD design (SOLIDWORKS)

 Structural and Thermal FE model (GMM and TMM) building

• FEM analysis (NASTRAN, Thermica, SOLIDWORKS Simulation)







Machining



- Internal or external suppliers
- 3, 4 and 5 axis machines
- TRL9
- Machining dimensions up to 6000x4500x1500 mm
- Tolerance up to 0.005 mm











Surface treatment



Conversion coatings

Types trivalent chromium coating (SURTEC650) and hexavalent chromium coating (Alodine 1200)

Bath dimensions 950 x 500 x 180 mm

Alloys to be coated 1xxx, 2xxx, 5xxx, 6xxx and 7xxx series of aluminum alloys

Technology manual, not automatic

Properties

Colour pale or dark grey Coating weight $0.1 - 0.5 \text{ g/m}^2$

Qualification

Corrosion 168h / 72h NSS and 240h humidity test

Bake-out 72h, 60° C, 10^{-5} mbar TVC 100 cycles $\pm 100^{\circ}$ C

Compliance

Standard ECSS-Q-ST-70-14C, prEN4729 and SP-ADST-1000112306 by Airbus

Acceptance by ESA and Airbus

Reference

Project Sentinel-2 flight hardware accepted by Airbus and ESA, CHEOPS, JUICE



Painting



Paint systems

Hardware size max. 2m²

Painting booth ISO Class 8 grade

Masking yes

Paints MAP PU1, MAP SG121FD, MAP PUK, Aeroglaze Z306

AQ PUK is under qualification

Primer yes but not necessary
Technology manual, not automatic

Overpainting within few hours of chromating

Curing in ISO Class 8 clean room or fast curing

Qualification

Corrosion 240h humidity test Bake-out 72h, 60° C, 10^{-5} mbar TVC 100 cycles $\pm 100^{\circ}$ C

Adhesion cross-cut and peel test

Compliance

Acceptance by ESA and Airbus



MLI



Performance: normal & high-efficiency

Appication: internal and external blankets

PROCESSES:

Manufacturing: manual or CNC cutting,

custom perforation

Assembly: blanket pinning,

blanket-to-blanket bonding by PSA,

Velcro application by PSA,

thermo-optical tape application

Grounding: grounding with Al foil and

removable fasteners or fixed rivet

Attachment: standoff or Velcro











Verification



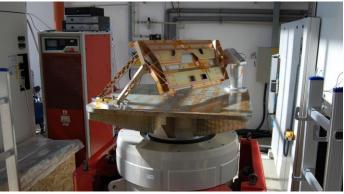
- CMM with 1800x800mm granite table. Accuracy is ~5micron.
- Portable Measuring Arm with ~1800mm volume.
 Accuracy is ~27microns.
- LDS shaker 35kN with slip table and head expanders (external)
- Surface Roughness measurement
- Detailed visual inspections by optical microscope
- Cleanliness measurement by portable particle counter
- Bake-out, TVAC and thermal balance
 - internal -70°C +100 °C
 - external AAC and ESA

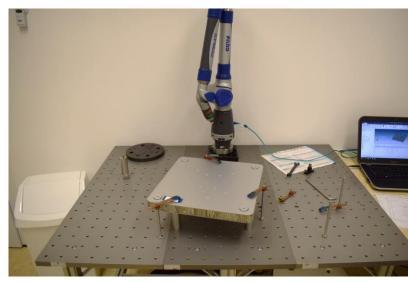












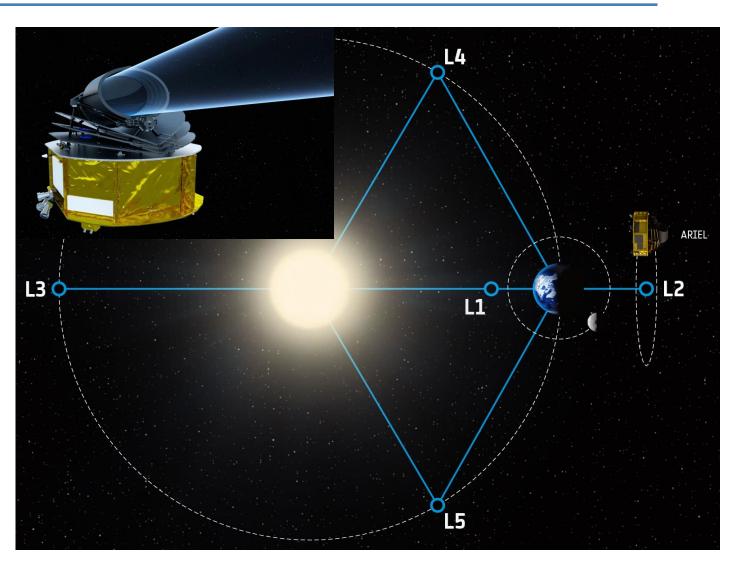


ARIEL - Overview



The mission:

- Elliptical primary mirror: 1.1 x 0.7 metres
- Mission lifetime: at least 4 years in orbit
- Payload mass / launch mass: ~500
 kg / ~ 1500kg
- Instrumentation: 3 photometric channels and 3 spectrometers covering continuously from 0.5 to 7.8 microns in wavelength
- Launch date: 2029
- Destination: Sun Earth Lagrange
 Point 2 (L2)
- Launch vehicle: Ariane 6-2. Launch shared with Comet Interceptor.



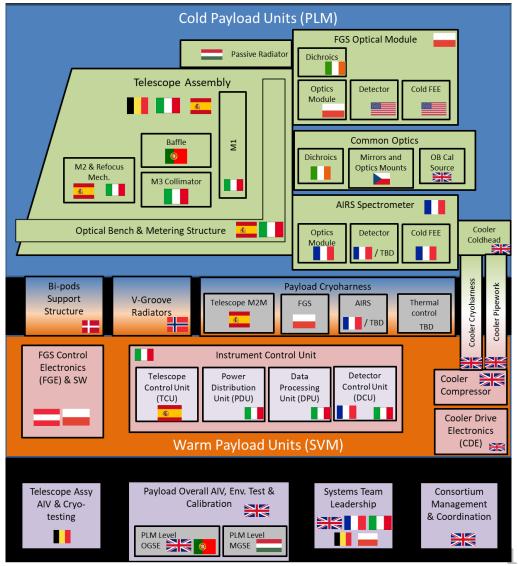
ARIEL - Overview



The Team:

- Ariel mission consortium is led by RAL space UK.
- Prime contractor is ADS
- Contributions from 22 ESA states and NASA and JAXA.



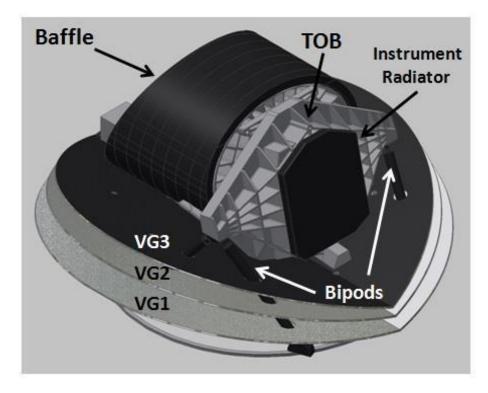


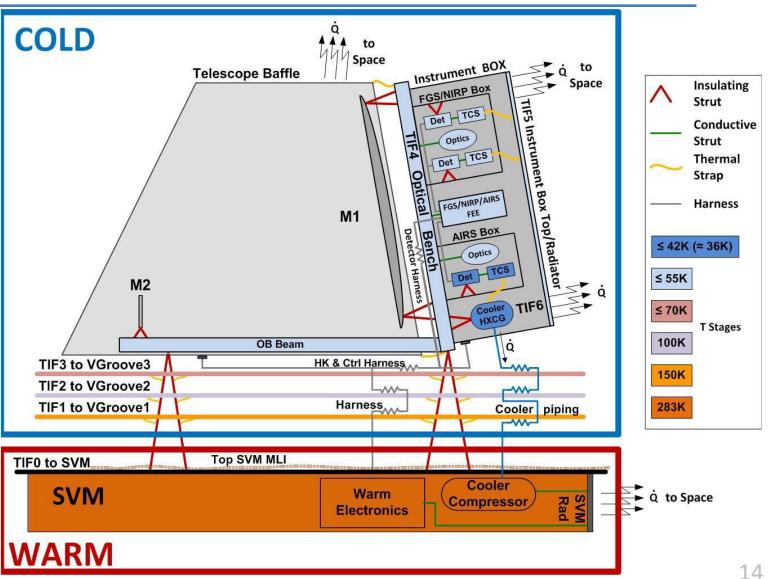
ARIEL - Payload



Purpose of Instrument Radiator:

- Instrument Cooling
- Radiation Shielding
- (TOB assembly stiffening)



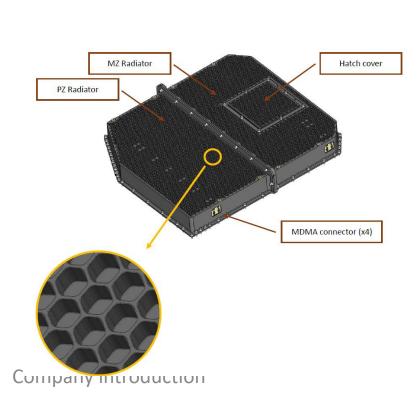


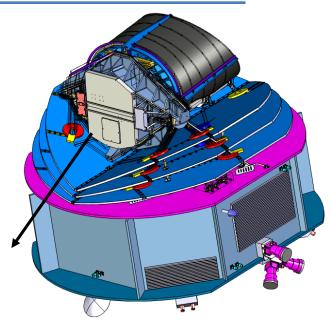
ARIEL – Flight hardware



Instrument Radiator Subsystem

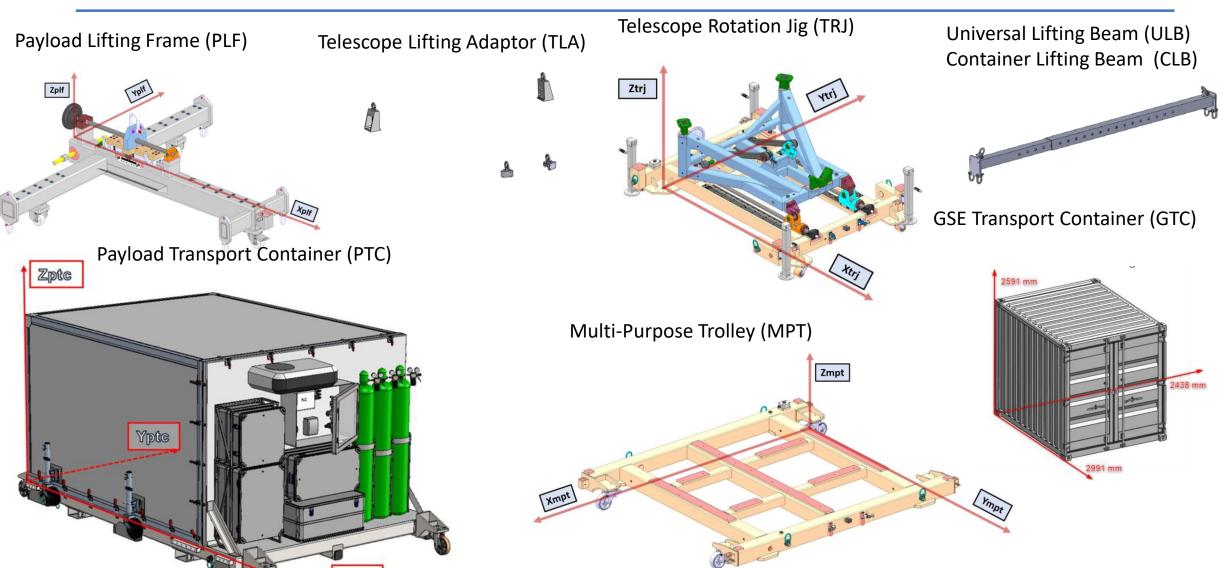
- Light tightly seals instrument cavity while ensuring the necessary radiative cooling capacity to dissipate heat produced by instruments into deep space.
- Challenging due to cryogenic (45K) operation temperature.
- Equipped with monitoring thermal sensors.
- Features special honeycomb radiative interface to compensate coating's low emittance at cryo.
- Build-to-spec





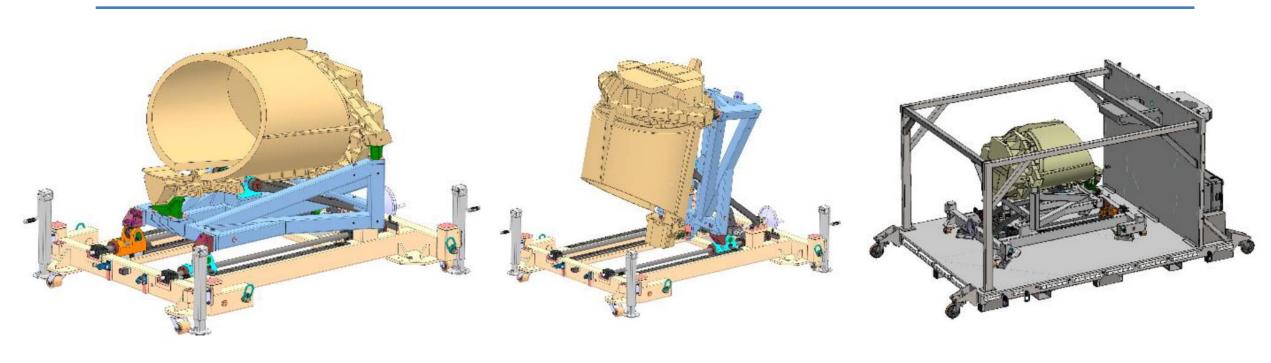
ARIEL – MGSE





ARIEL – MGSE





54. Figure – TRJ with TA

55. Figure – TRJ in integration position

TRJ and TA in the PTC

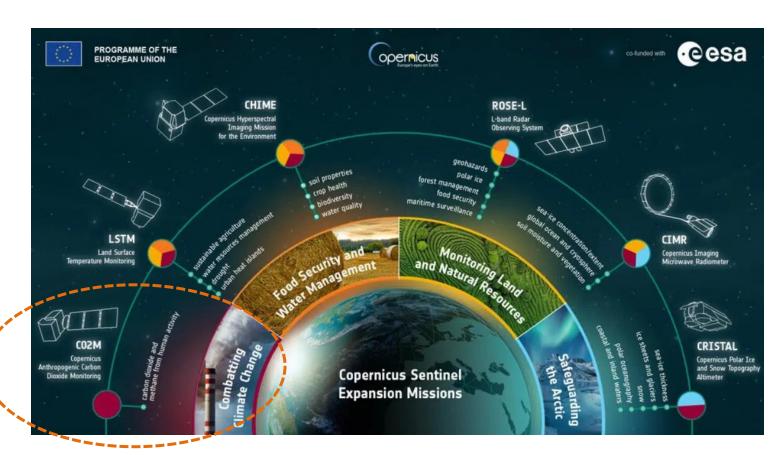
CO2M – Thermal Guard Assembly



Copernicus is one of the largest European space program for Earth Orbservation.

CO2M is part of ESA HPCM (High Priority Candidate Missions) / Copernicus Expansion Mission.





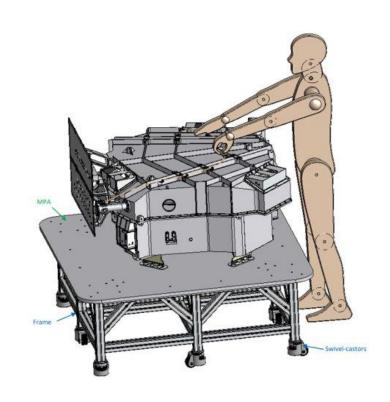
Mission is the climate change monitoring.

The aim of the CO2M mission is the detection and measurement of climate damaging gas focusing on carbon-dioxide and nitrogen-dioxide.



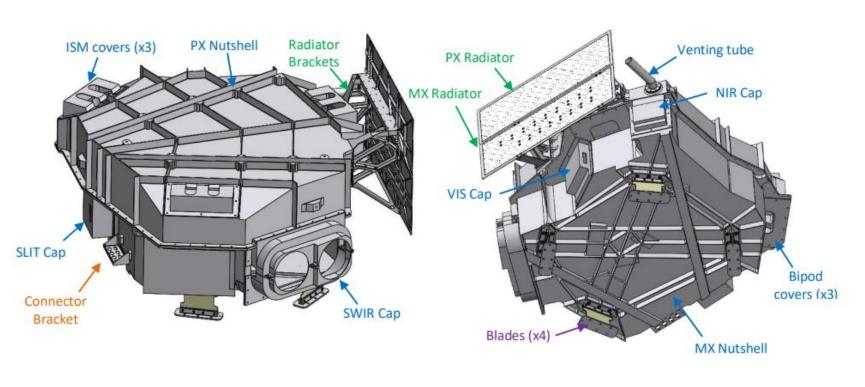


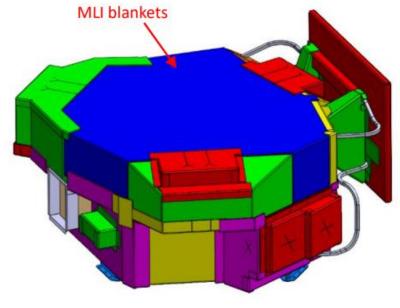
- Selection in open competition
- Encapsulation and thermal insulation of CO2 spectrometer.
- Design, analysis, manufacturing and testing of PFM, FM2 and FM3 (TBC)
- Platform provider: Thales Alenia
 Space France / Prime: OHB
- Build-to-spec





CO2M – Flight hardware





Comet Interceptor – Flight hardware

Target transfer (1-1.5 years)

Dimensions

Delta-V

Wet mass

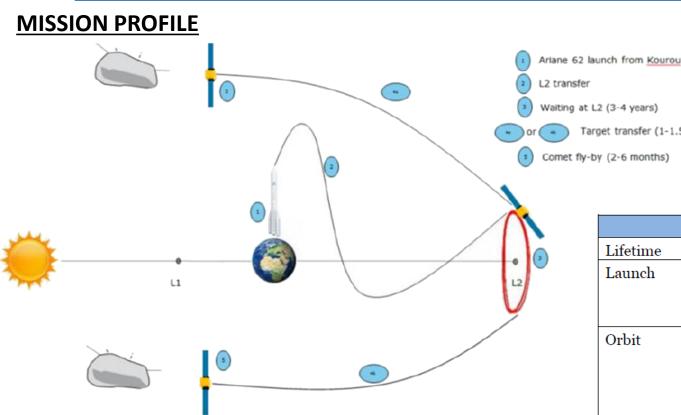
Deployed

Chemical propulsion

Electric propulsion

Stowed





The mission's primary science goal is to characterize, for the first time, a dynamically-new comet or interstellar object, including its surface composition, shape, and structure, the composition of its gas coma.

Prime: OHB Italy

CoCa leader: University of Bern

Comet Interceptor – Mission summary		
Lifetime	Nominal 5 years with maximum 6 months of Science Operations	
Launch	Launcher	Ariane 6.2
	Date	2028
	Configuration	Shared, dual launch with ARIEL mission
Orbit	Waiting phase	Halo orbit around SEL2 point
	Target transfer phase	Heliocentric trajectory close to Earth orbit
	Conditions at encounter	Heliocentric distances between 0.9 and 1.25 AU Solar phase angle range at encounter +/-45 deg Fly-by relative velocity at encounter ≤70 km/s
Overall system characteristics		
Mass	Dry mass	655 kg
1		

110 m/s

1522.5 m/s

796 kg (incl. probes B1 and B2)

1,974 mm x 2,073 mm x 1,976 mm

9,768 mm x 2,999 mm x 2,484 mm

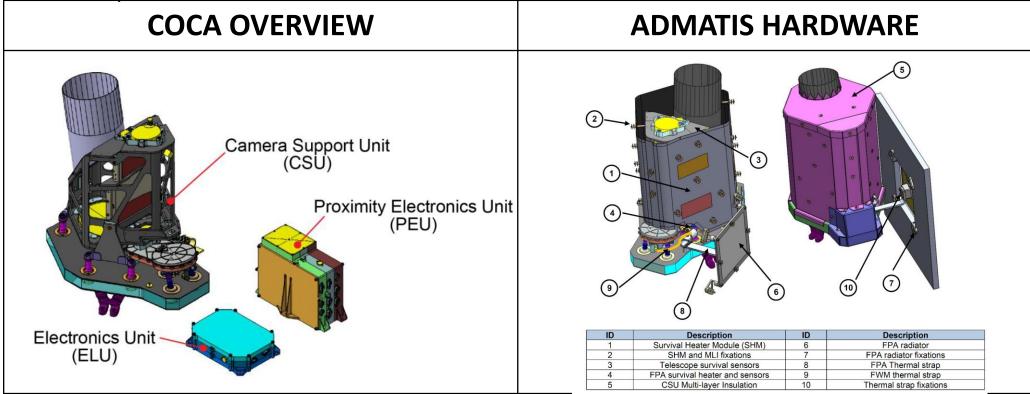
COMET



* Admatis Advanced Materials In Space

- Comet Interceptor Flight hardware
- ESA Project
- Objective is to develop passive and active thermal control hardware for Comet Interceptor's instrument called COmet Camera (CoCa).
- Active and Passive Thermal Control packages are under ADMATIS responsibilities.

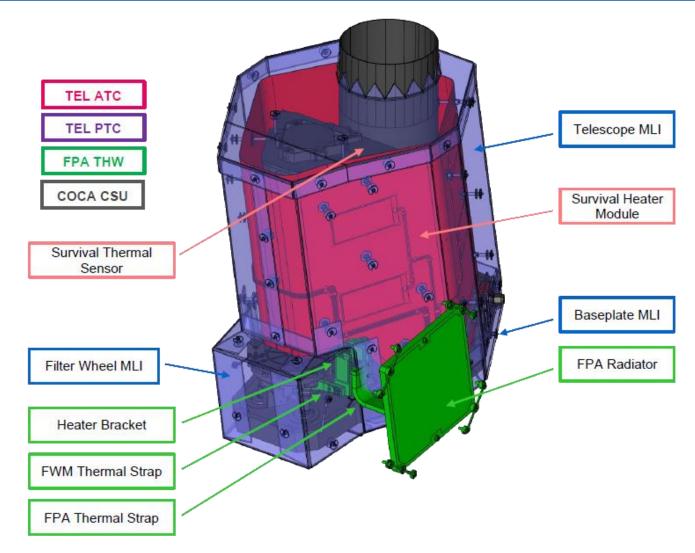
Build-to-spec





23

Comet Interceptor – Flight hardware

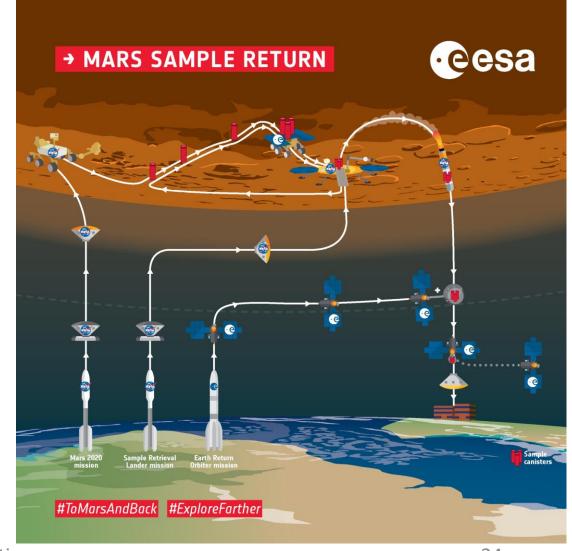




*** Admatis Advanced Materials In Space

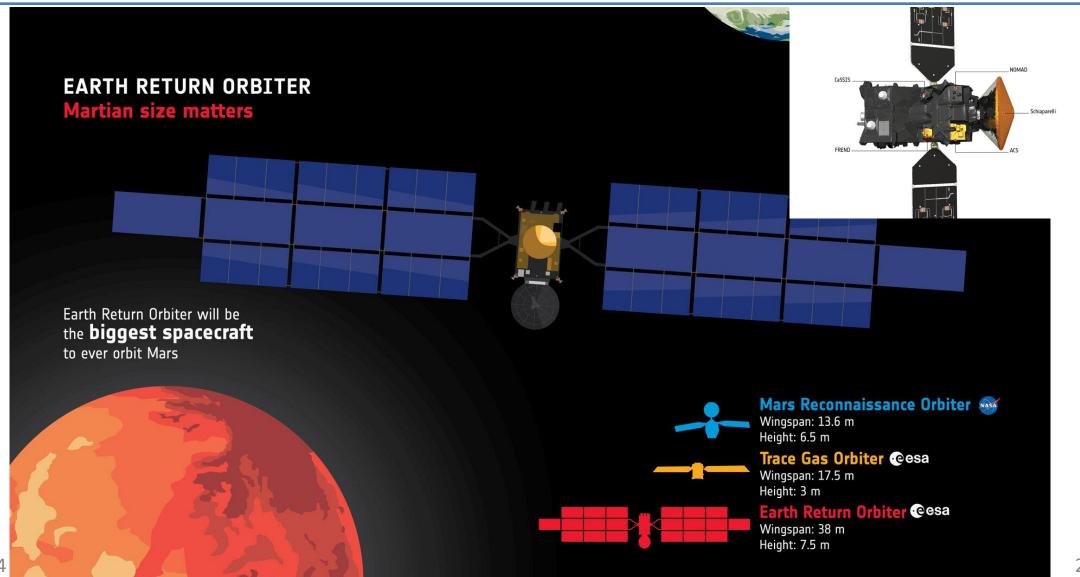
MSR – ERO –SDS – The Mission

- External scientific instrument for radiation measurement
- Prime: Airbus
- Build-to-spec
- 2022 2025
- Hungarian team: EK, ADMATIS, Remred
- ADMATIS is responsible for thermal and structural subsystems



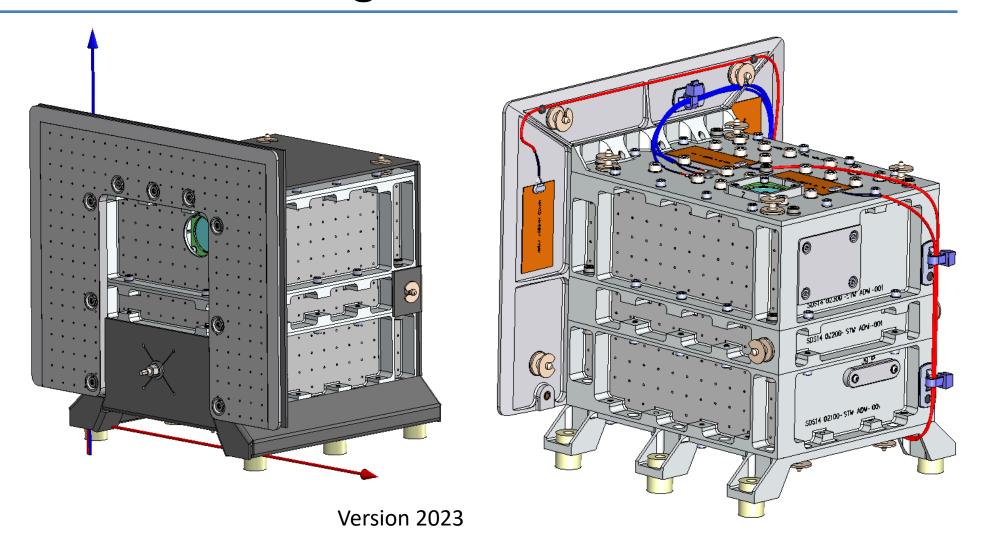


MSR – ERO –SDS – Spacecraft





MSR – ERO –SDS – Flight hardware



Mission Overview - MSN



ESA CLEANSPACE

CLEAN SPACE IS THE EUROPEAN SPACE AGENCY'S INITIATIVE, STARTED IN 2009 WITH THE ECOSAT STUDY, TO SAFEGUARD THE TERRESTRIAL AND ORBITAL ENVIRONMENTS, WHILE BOOSTING THE INNOVATION AND COMPETITIVENESS OF EUROPE'S SPACE SECTOR.

CLEAN SPACE HAS THREE BRANCHES:

They reflect its mission to assess the environmental impacts of Agency programmes designed to find ways to address these challenges and contribute to a more sustainable and competitive European space industry.

- **ECODESIGN**: Embedding environmental sustainability within space mission design.
- MANAGEMENT OF END OF LIFE: Developing technologies to prevent the creation of future debris.
- **IN-ORBIT SERVICING**: Actively removing spacecraft from orbit and demonstrating in-orbit servicing.



Clean Space (esa.int)

Markers Supporting Navigation

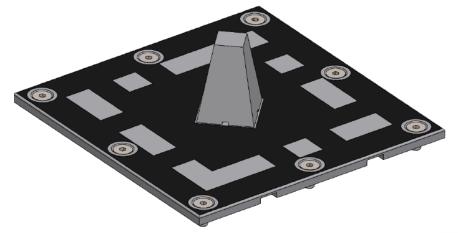


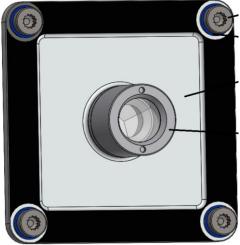
ESA R&D project portfolio with 5 projects at ADMATIS.

The first was competitive with other possible solutions.

Objectives:

- develop something than can be detected after 15 years LEO contitions with visible and thermal infrared cameras
- BOL and EOL properties are almost the same
- support navigation with range 40 0 meter
- detectable from Earth by laser technology





Markers Supporting Navigation



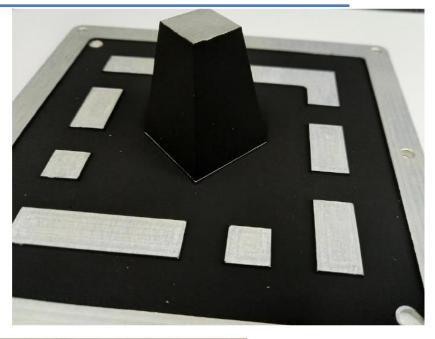
ESA – ADMATIS exclusive supplier contract signed in 2023.

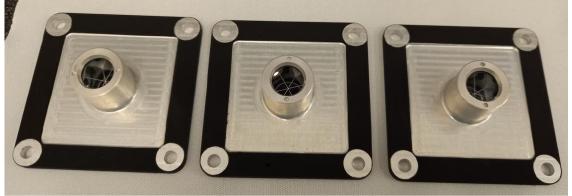
The first industrial orders received:

- ESA AVS (AVS)
- ESA HPCM CRISTAL (Airbus)
- ESA HPCM LSTM (Airbus)

New developments running:

- phosphorescent painted Markers
- GNC test facility
- Markers for constellations
- System level leader of international team



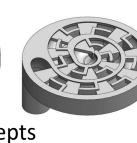


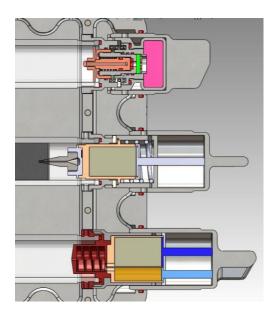
POSITIVE – Flight hardware

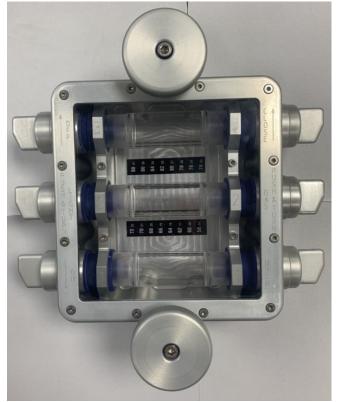


- Porous Lightweight Material Development
- Part of Hungarian in Orbit (HUNOR)
- Objective is to develop technologies for building structural porous material from Moon dust in order to support colonisation. Two types of foam are under development:
 - High porosity insulation material
 - Low porosity structural material like a brick
- **Experiments on ISS**
- 2022 2025









References





Metallic Mechanical and Thermal Hardware are designed,

manufactured, tested and delivered by ADMATIS.

Partner: Airbus

Implementation: 2009-2018



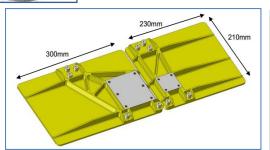
FPA and FEE radiator development to CHEOPS

satellite.

Partner: ESA, University of Bern

Implementation: 2013-2017





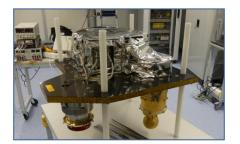


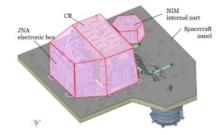
JUICE

STM and FH MLI for PEP and Thermal dummies for SWI instrument

Partner: ESA, Airbus, University of Bern

Implementation: 2020





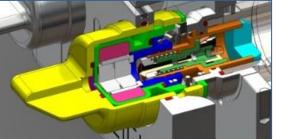
FOCUS

Foaming experiments on board of ISS in February 2010.

Scientific background and hardware development.

Partner: European Space Agency (ESA)

Implementation: 2007-2010









Facilities



Software

- servers with Linux and Windows
- 2. laptops with Windows, MS Office and ESET
- 3. SolidWorks Premium 3D CAD
- 4. SolidWorks Simulation Premium
- 5. NASTRAN
- 6. Thermica
- 7. LabView
- 8. MS SharePoint
- 9. Microsoft Project

Premises

- 1. Clean rooms for manufacturing and assembly (ISO8)
- 2. Clean room for cleaning and packaging (ISO8)
- 3. Surface treatment line and chemical laboratory
- 4. Conference rooms equipped with teleconf
- 5. Storage rooms with controlled environment

Equipment

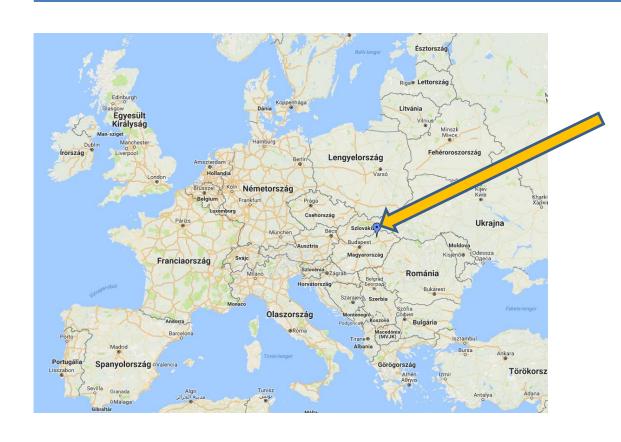
- Clean Bench (ISO5)
- 2. Surface treatment line for SURTEC and Alodine
- 3. Painting booth
- 4. Curing booth
- 5. MLI assembly area
- 6. TVC for bake-out, thermal cycling and thermal balance tests
- 7. Thermal cycling
- 8. CMMs for 3D measurement (portable arm, bridge)
- 9. NSS chamber
- 10. Humidity chamber
- 11. Thermal imaging
- 12. Microscope
- 13. Analytical scale
- 14. CNC mill for MLI cutting

At our partners (HUNSPACE)

- 1. Vibration test house (University of Dunaújváros)
- 2. Mechanical test house (University of Miskolc)
- 3. Metal machining (several)

Contact





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